A Self-Coherence Based Anti-Jamming GPS Receiver

Moeness G. Amin

Center for Advanced Communications College of Engineering Villanova University Villanova, PA 19085-1699 Moeness.amin@villanova.edu

Sponsored by ONR and AFRL

3/10/05

OUTLINE

GPS overview

- Existing techniques in GPS interference suppression
- Self-coherence based anti-jamming GPS receiver
- Simulation results
- Conclusions

WHAT IS GPS?



Moeness Amin Center for Advanced Communications

GPS is a satellitebased all-weather navigation system providing precise position, velocity, and timing information. Global navigation systems

- GPS: US
- GLONASS: Russia
- Galileo: Europe

GPS COMPONENTS



Moeness Amin Center for Advanced Communications Space segment: satellite constellation
 Control segment: ground stations
 User segment: receivers

HOW GPS WORKS?

GPS uses one-way time-ofarrival ranging to determine user position - measure the direct path signal travel time from a satellites to a user's receiving device.



 – Pseudorange: the range from the satellite plus clock offset



Geolocation of the receiver: the intersection of pseudoranges from a set of satellites (minimum 4).

GPS SIGNAL

GPS employs direct-sequence spreadspectrum (DS-SS) signaling



Two L-band frequencies: – L1=1.57542 GHz – L2=1.2276 GHz

Two pseudorandom codes

Coarse/acquisition (C/A) code: chip rate 1.023
 Mchips/sec, BW 2 MHz, repeats every millisec.

Moeness Amin Center for Advanced Communications – Precision (P) code: chip rate 10.23 Mchips/sec, BW 20 MHz, repeats about every week (military use).

GPS SIGNAL (Cont'd)

GPS C/A signal waveform:



CHALLENGES IN GPS

Two dominant sources of errors in precision GPS:

 Interference: reduces the SNR of the GPS signal such that the receiver is unable to obtain measurements from the GPS satellite.

 Multipath: broadens and biases the crosscorrelation function.

GPS AND DSSS SYSTEMS

Common

- Both use PN
- Both require synchronization
 - DSSS: despreading
 - GPS: despreading and pseudorange measurement

Difference

- Multipath
 - DSSS: improves system performance due to diversity – constructive ©
 - ◆ GPS: results in erroneous pseudorange measurements – destructive ☺
- Near-far phenomena
 - ◆ DSSS: a severe problem ☺
 - GPS: not a problem [©]

INTERFERENCE IN GPS

Types of Interference in GPS:
 Additive white Gaussian noise (AWGN)
 Broadband interference can sometimes be

modeled as AWGN

Continuous wave (CW)
 A pure tone or narrowband modulated signal.
 Chirp signal

Pulsed interference (e.g., radars)
 – Effectively "shoots holes" in the received signal

EXISTING TECHNIQUES

Maximum Likelihood Approach

(Time-delay and carrier-phase estimation)

- Single-antenna approach
- Multiple-antenna approach

Suboptimum Methods

(Interference Suppression)

- Space-time adaptive method
- Time-frequency method
- Navigation data demodulation

EXISTING TECHNIQUES

GPS interference suppression

Requires satellite locations • MMSE • MSINR A • ML

Moeness Amin Center for Advanced Communications Assumes bit synchronization

 Spatial-temporal matched filtering Requires neither • Self-coherence based antijamming approach

Interference Representation

Time-frequency domain



INTERFERENCE SUPPRESSION IN GPS (Cont'd)

Spread-spectrum (SS) provides certain protections against interference

 Receiver fails when the interference to signal ratio exceeds the 30 dB.

Interference suppression
 Separate domain: time, space, or frequency
 Joint domain: time-frequency or space-time

Moeness Amin Center for Advanced Communications None of these methods fully utilizes the repetitive feature of the GPS C/A-code.

GPS SIGNAL STRUCTURE

Noise-free GPS data structure



Two blocks of data (N consecutive samples each)

Data block

Moeness Amin Center for Advanced Communications Reference block: *jP* samples apart from data block

The NOTION OF SELF-COHERENCE

Self-coherent signal

The correlation between the signal and its frequency-shifted version is nonzero for some time lag.

Significance of self-coherence

 Blindly extract the desired signals in the presence of unknown noise and interference.
 SCORE algorithm

SELF-COHERENCE (Cont'd)

Example:

x(t) = as(t) + v(t)

- s(t): self-coherent signal [not v(t)] Cyclic autocorrelation of x(t)

 $R_{xx}^{(\beta)}(\tau) = |a|^2 R_{ss}^{(\beta)}(\tau) + R_{vv}^{(\beta)}(\tau) = |a|^2 R_{ss}^{(\beta)}(\tau)$ Frequency shift completely decorrelates the interference component in *x*(*t*).

SELF-COHERENCE GPS RECEIVER

Receiver structure



M-antenna array
 Two
 beamformers
 - w: generates
 output signal
 - f: generates
 reference
 signal

SELF-COHERENCE GPS RECEIVER (Cont'd)

Assumptions:

- Data block and reference blocks are within the same navigation symbol.
- Interference does not have the same periodic structure as that of the GPS C/A signals.
- The proposed receiver acts like a preprocessor to suppress interference of all satellites.
 - Conventional multipath mitigation techniques, such as delay lock loop (DLL), can then operate on significantly higher SINR than that encountered at the receiver input.

RECEIVED SIGNAL

Received signal at the chip-rate in the data block

$$\mathbf{x}(n) = \sum_{q=0}^{Q} S_q(n - \tau_q) \mathbf{a}_q e^{j\phi_q} + u(n)\mathbf{d} + \mathbf{v}(n)$$

S₀(n) is direct-path signal
 If the GPS signal, interference, and noise are independent

$$\mathbf{R}_{xx} = E\left\{\mathbf{x}(n)\mathbf{x}^{H}(n)\right\} = \mathbf{R}_{s} + \mathbf{R}_{u} + \mathbf{R}_{v}$$

RECEIVED SIGNAL (Cont'd)

The samples of GPS signal in the reference block have the same values as the corresponding samples in the data block within the same symbol

 $\mathbf{x}(n) = \mathbf{x}(n-jP), \ 1 \le j < 20$

If the GPS signals are the only data components that are correlated when delayed *jP* samples $\mathbf{R}_{xx}^{(P)} = E \{ \mathbf{x}(n) \mathbf{x}^{H} (n - jP) \} = \mathbf{R}_{s}$

PROPOSED RECEIVER
Cost function – SCORE algorithm

$$C(\mathbf{w}, \mathbf{f}) = \frac{\left|R_{zd}\right|^2}{R_{zz}R_{dd}} = \frac{\left|\mathbf{w}^H \mathbf{R}_{xx}^{(P)} \mathbf{f}\right|}{\left[\mathbf{w}^H \mathbf{R}_{xx} \mathbf{w}\right] \left[\mathbf{f}^H \mathbf{R}_{xx} \mathbf{f}\right]}$$

w is obtained by maximizing C(w, f)

- Beamformer output: $z(n) = \mathbf{w}^H \mathbf{x}(n)$

- Reference signal: $d(n) = \mathbf{f}^H \mathbf{x}(n - jP)$

- Cross-correlation between z(n) and d(n):

$$R_{zd} = E\left\{z(n)d^H(n)\right\}$$

PROPOSED RECEIVER (Cont'd)

Error signal

e(n) = z(n) - d(n)

Least-squares solution of f

 $\mathbf{f}_{LS} = \mathbf{R}_{xx}^{-1} \mathbf{R}_{xx}^{(P)H} \mathbf{w}$

The weight vector w that maximizes the cost function is the eigenvector corresponding to the largest eigenvalue of the generalized eigenvalue problem

$$\mathbf{R}_{xx}\mathbf{w} = \lambda_{\max}\mathbf{R}_{xx}^{(P)}\mathbf{R}_{xx}^{-1}\mathbf{R}_{xx}^{(P)}\mathbf{w}$$

Exact Expression

Define events

 $A_1: x(n) \& x(n-jP)$ are within the same symbol, $A_{21}: x(n) \& x(n-jP)$ are in two symbolswith the same sign, $A_{22}: x(n) \& x(n-jP)$ are in two symbolswith different signs

The corresponding probabilities are $\Pr\{A_1\} = 1 - \frac{jP}{20P}$ $\Pr\{A_{21}\} = \Pr\{A_{22}\} = \frac{jP}{40P}$

Continue

$$\mathbf{R}_{xx}^{(P)} = E\left\{\mathbf{x}(n)\mathbf{x}^{H}(n-jP)\middle|A_{1}\right\} \Pr\left\{A_{1}\right\}$$
$$+ E\left\{\mathbf{x}(n)\mathbf{x}^{H}(n-jP)\middle|A_{21}\right\} \Pr\left\{A_{21}\right\}$$
$$+ E\left\{\mathbf{x}(n)\mathbf{x}^{H}(n-jP)\middle|A_{21}\right\} \Pr\left\{A_{21}\right\}$$
$$= \left(1 - \frac{jP}{20P}\right)\mathbf{R}_{s}$$

The cross-SCORE based receiver does not have the ability to mitigate multipath.
 Since multipath often comes from near the horizon while the GPS satellites are located above the horizon, adding constraints to the previous receiver can mitigate multipath entering the receiver from near the horizon.

Define the matrix containing steering vectors associated equally spaced directions covering solid angle Ω near the horizon

 $\mathbf{B} = \begin{bmatrix} \mathbf{b}(\gamma_1) & \cdots & \mathbf{b}(\gamma_D) \end{bmatrix}$ The modified cost function on f $\mathbf{f}_{opt} = \arg \max_{\mathbf{f}} \frac{\mathbf{f}^H \tilde{\mathbf{R}}_{xx} \mathbf{f}}{\mathbf{f}^H \mathbf{R}_{yy} \mathbf{f}}, \text{ subject to } \mathbf{B}^H \mathbf{f} = \mathbf{0}$

 $\tilde{\mathbf{R}}_{rr} = \mathbf{R}_{rr}^{(P)} \mathbf{R}_{rr}^{-1} \mathbf{R}_{rr}^{(P)H}$

• Let A be the matrix that spans the null space of B such that $B^H A = 0$.

Let α be a vector such that f = Aα.
 Using the vector α, the constrained maximization problem can be transformed into the unconstrained maximization problem as

 $\boldsymbol{\alpha} = \arg \max_{\boldsymbol{\alpha}} \frac{\boldsymbol{\alpha}^{H} \mathbf{A}^{H} \mathbf{\tilde{R}}_{xx} \mathbf{A} \boldsymbol{\alpha}}{\boldsymbol{\alpha}^{H} \mathbf{A}^{H} \mathbf{R}_{xx} \mathbf{A} \boldsymbol{\alpha}}$

Solving the above unconstrained generalized eigenvalue problem lead to α which is given by the eigenvector associated with the largest eigenvalue. The beamformer f_{opt} be obtained correspondingly.

The beamformer w_{opt} is then given by

 $\mathbf{W}_{opt} = \mathbf{R}_{xx}^{-1} \mathbf{R}_{xx}^{(P)} \mathbf{f}_{opt} = \mathbf{R}_{xx}^{-1} \mathbf{R}_{xx}^{(P)} \mathbf{A}\boldsymbol{\alpha}$

IMPLEMENTATION ISSUES

Sample estimates have to be used instead of the exact ones

$$\hat{\mathbf{R}}_{xx} = \frac{1}{N} \mathbf{X}_N \mathbf{X}_N^H, \quad \hat{\mathbf{R}}_{xx}^{(P)} = \frac{1}{N} \mathbf{X}_N \mathbf{X}_{Nref}^H$$

Data sample matrix

$$\mathbf{X}_{N} = \begin{bmatrix} \mathbf{x}(n), & \cdots, & \mathbf{x}(n - (N - 1)) \end{bmatrix}$$

Reference sample matrix

$$\mathbf{X}_{Nref} = \begin{bmatrix} \mathbf{x}(n-jP), & \cdots, & \mathbf{x}(n-(N-1)-jP) \end{bmatrix}$$

IMPLEMENTATION ISSUES (Cont'd)

Multiple data and reference blocks can be used to improve the time-averaging:



$$\hat{\mathbf{R}}_{xx} = \frac{1}{G} \sum_{g=1}^{G} \mathbf{X}_{N}(g) \mathbf{X}_{N}^{H}(g) / N$$
$$\hat{\mathbf{R}}_{xx}^{(P)} = \frac{1}{G} \sum_{g=1}^{G} \mathbf{X}_{N}(g) \mathbf{X}_{Nref}^{H}(g) / N$$

COVARIANCE ESTIMATION (Cont'd)

The expected values of estimates

$$\bar{\hat{\mathbf{R}}}_{xx}^{(P)} = \left(1 - \frac{jP}{20P}\right) \mathbf{R}_s, \text{ one block estimation}$$
$$\bar{\hat{\mathbf{R}}}_{xxG}^{(P)} = \left(1 - \frac{jP}{20P}\right) \mathbf{R}_s, \mathbf{G} \text{ blocks estimation}$$

The corresponding variances of estimates $\operatorname{var}\left\{\hat{\mathbf{R}}_{xx}^{(P)}\right\} = \frac{M(N + \sigma_{v}^{2})}{N} \mathbf{R}_{s} + \frac{M(1 + \sigma_{v}^{2})}{N} \mathbf{R}_{v} - \left(\overline{\hat{\mathbf{R}}}_{xx}^{(P)}\right)^{2}$ $\operatorname{var}\left\{\hat{\mathbf{R}}_{xxG}^{(P)}\right\} = \left(1 - 2\frac{jP}{20P} + 2\frac{jP}{20GP}\right) \left[\frac{M(N + \sigma_{v}^{2})}{N} \mathbf{R}_{s} + \frac{M(1 + \sigma_{v}^{2})}{N} \mathbf{R}_{v}\right] - \left(\overline{\hat{\mathbf{R}}}_{xxG}^{(P)}\right)^{2}$

SIMULATION RESULTS

 Linear uniform array with *M*=7 sensors.
 GPS navigation symbols in BPSK format; C/A-code with processing gain of *P*=1023.
 N=800 samples in both the data and reference blocks.

Jammers are generated as broadband binary signals with the same rate as the C/A-code.

Multipath signal power is one-fifth of the direct-path signal power.

Critical Case

Beam pattern without interference – evenly split data block (G=2, SNR=-30 dB)





IMPROVED!

Beam pattern without interference (G=7, SNR=-30 dB)





FOR ALL SATELLITES

With multiple satellites in the field of view (M=9, SNR=-30 dB)



SIMULATION RESULTS (Cont'd)

Comparison with MMSE method (spatial processing only (SINR=-33 dB, JSR=30 dB)



Performance Circular array: *M*=7 sensors, SNR=-30 dB and JSR=30 dB.



Jammers with Similar Coherence

Jammers with the same structure as the GPS signals (satellite at 30 degree and two jammers at 10 and 50 degree)



MULTIPATH!

In the presence of multipath (satellite at 30 degree, one multipath at 8 degrees)



SIMULATION RESULTS (Cont'd)

Comparison between the original and modified approaches in the presence of multipath (10 degree, SNR = -30 dB, 1/5 power)







Wei Sun Center for Advanced Communications







CONCLUSIONS

- Presented a novel self-coherence based GPS anti-jam receiver.
- The receiver utilizes the inherent selfcoherence feature of the GPS C/A signal.
- The proposed receiver requires neither the knowledge of the transmitted GPS signal nor the location of the satellite.

The proposed receiver is able to suppress a large class of interference as long as the interferers do not have the same periodic structure as the C/A signal.

REFERENCE

[Braash99] M. Braasch and A. Dierendonck, "GPS receiver architectures and measurements," *Proceedings of the IEEE*, vol. 87, no. 1, pp. 48–64, January 1999.

[Kaplan96] E. Kaplan (eds.), *Understanding GPS: Principles and Applications*. Artech House Publisher, 1996.

[Xiong03] P. Xiong, M. Medley, and S. Batalama, "Spatial and temporal processing for global navigation satellite systems: The GPS receiver paradigm," *IEEE Transactions on Aerospace and Electronic Systems*, vol. 39, no. 4, pp. 1471–1484, October 2003.

[Agee90] B. Agee, S. Schell, and W. Gardner, "Spectral selfcoherence restoral: A new approach to blind adaptive signal extraction using antenna array," *Proceedings of the IEEE*, vol. 78, no. 4, pp. 753–767, April 1990.

REFERENCE

[Fante00] R. Fante and J. Vaccaro, "Wideband cancellation of interference in a GPS receive array," *IEEE Transactions on Aerospace and Electronic Systems*, vol. 36, no. 2, pp. 549–564, April 2000.

[Seco98] G. Seco and J. Fernandez-Rubio, "Array signal processing techniques for pseudorange and carrier phase measurement," in *Proceedings of the 2nd European Symposium on Global Navigation Satellite Systems (GNSS98)*, vol. IX-P-10, 1998, pp. 20–23.

[Amin04] M. Amin, L. Zhao, and A. Lindsey, "Subspace array processing for the suppression of FM jamming in GPS receivers," *IEEE Transactions on Aerospace and Electronic Systems*, to appear.